

19 of these alternative approaches to steel structure design.

20 **Keywords:** Buildings (codes); design (buildings); inelastic analysis; reliability; steel; struc-
21 tural engineering

22 INTRODUCTION

23 Appendix 1 - Inelastic Analysis and Design - of the American Steel Specification [AISC](#)
24 [360-10 \(2010\)](#) permits, for the first time, the use of inelastic analysis of overall system re-
25 sponse to design a steel frame. The inelastic analysis must take into account geometric
26 and material nonlinearity, including the effect of residual stress and geometric imperfections.
27 Appendix 1 states that “Strength limit states detected by an inelastic analysis that incor-
28 porates all sources of major nonlinear actions are not subject to the LRFD member-based
29 design checks in the Specification when a comparable or higher level of reliability is provided
30 by the analysis.” Appendix 1 represents a new system-based paradigm for steel frame de-
31 sign, and offers several important advantages over the existing LRFD method. A rigorous
32 inelastic analysis can accurately determine the complex interactions between members of a
33 large structural system, and capture the beneficial system effect of load redistribution after
34 the initial formation of plastic hinges ([Ziemian et al., 1991](#); [Clarke et al., 1992](#); [Chen and](#)
35 [Kim, 1997](#); [White and Hajjar, 2000](#); [Kim et al., 2001](#); [Trahair and Chan, 2003](#); [Ngo-Huu](#)
36 [et al., 2007](#); [Zhang and Rasmussen, 2013](#); [Zhang et al., 2014](#)). Thus, inelastic analysis often
37 leads to the design of lighter and more economic structures than LRFD, which is based on a
38 “first-hinge” approach. Inelastic analyses can determine structural performance from initial
39 loading up to collapse, whereby designers are better able to understand the system behavior.
40 This feature is especially important in the new paradigm of performance-based design which
41 is closely coupled to the issue of system behavior. Since inelastic analysis can explicitly in-
42 dicate the limit states (e.g., onset of structural nonlinearity, incipient system collapse, etc.),
43 it becomes possible to identify different performance limits in design.

44 The inelastic method is different from the “Direct Analysis Method” (abbreviated as
45 “DM” in the *AISC Specification*) as it appears in Chapter C of *AISC 360-10*. The DM

46 is based on a rigorous second-order elastic analysis that directly models member imper-
47 fections, and uses reduced member stiffnesses to account for the influence of inelasticity
48 (including residual stress) on the stability of structures. While member and system stabil-
49 ity are checked/detected by the analysis, the equation based design checks only need to be
50 completed at the cross section level, as stipulated in the remaining Chapters of the *Spec-*
51 *ification*. While the DM eliminates the need for calculating effective length factors, it is
52 still a member-based design approach as opposed to the system-level checks in the inelastic
53 method. The inelastic method is an extension of the DM.

54 From a structural analysis point of view, the technical barriers to the use of inelastic
55 method in practical design have diminished, as significant advances in computerized struc-
56 tural analysis have occurred during the last two decades. Structural analysis software used by
57 structural engineers nowadays often incorporates various levels of inelastic analysis (plastic-
58 hinge analysis or plastic-zone analysis). Unfortunately, even with the advanced nonlinear
59 structural analysis method, the actual performance of a steel frame cannot be predicted with
60 certainty because uncertainties in structural loads, material strength and stiffness will always
61 be present. Because of this, Appendix 1 of *AISC 360-10* requires that the inelastic analysis
62 must take into account the uncertainties in system, member, and connection strength and
63 stiffness. According to Section 1.3.1 of Appendix 1, the acceptable method for including
64 uncertainty in system, member, and connection strength and stiffness is to reduce the yield
65 strength and the stiffness of all steel members and connections by a factor of 0.90 for the
66 inelastic analysis. The *AISC 360-10 Commentary* acknowledges that this reduction factor
67 of 0.90 has its origin in the AISC LRFD resistance factors for tension and flexural members
68 governed by the yield limit states; its use in system-based design, although “deemed accept-
69 able”, is not based on any system reliability calibration. An important question thus has
70 been raised: is the current reduction factor of 0.90 sufficient to fulfill the goal of assuring an
71 “acceptable” level of structural reliability in system-based design?

72 In the past decade, several research efforts have been made to investigate the system

73 reliabilities of steel frames and determine the resistance factors for the system-based design
74 by the inelastic method. [Buonopane and Shafer \(2006\)](#) compared the system reliabilities
75 of a group of sixteen, closely related planar low-rise steel frames designed by both LRFD
76 and inelastic analysis methods. The frames were subjected to gravity loads. For a target
77 reliability index of 3.0 on system strength, it was found that the values of system resistance
78 factor range from 0.86 to 0.91. This study only considered the uncertainties in the structural
79 loads and the yield strength. Other random effects, such as the uncertainties in the cross-
80 sectional properties, elastic modulus, and geometric imperfections, were ignored. This study
81 also used a coefficient of variation (COV) of 0.10 for live load, which underestimated the
82 variability of live load as various studies have shown that live load has a COV of about 0.25
83 ([Ellingwood et al., 1982](#)). Accordingly, the system resistance factors derived in [Buonopane
84 and Shafer \(2006\)](#) may be somewhat higher than warranted.

85 [Zhang et al. \(2014, 2016a,b\)](#) presented a framework for analysing the system reliability
86 of steel frames and determining the system resistance factors for the design-by-inelastic
87 method. The reliability framework is based on the First-Order Reliability Method (FORM).
88 The system resistance factors were calculated for a series of low- to mid-rise planar braced
89 and moment resisting frames for various target reliability levels. Different failure modes and
90 loading conditions (e.g., live-to-dead load ratios) were considered. All important random
91 variables were considered in the reliability assessments, including the uncertainties in yield
92 strength, modulus of elasticity, cross-sectional properties, residual stress, initial geometric
93 imperfections, and structural loads. The study showed that a system resistance factor of
94 0.80-0.85 would be required for the inelastic method to achieve a target system reliability
95 index of 3.0-3.25.

96 The steel frames studied in [Buonopane and Shafer \(2006\)](#), [Buonopane \(2008\)](#) and [Zhang
97 et al. \(2014, 2016a,b\)](#) all sustained significant yielding (formation of multiple plastic hinges)
98 when the frames were at the state of incipient collapse. For this category of structures, design
99 by inelastic methods can utilize the frame strength remaining after first yielding to sustain

100 further loading. The design strengths of such frames predicted by the inelastic analysis can be
101 10%-30% higher than those estimated from the current elastic-LRFD method. Studies have
102 shown that if such frames are designed by the inelastic method, their system reliability would
103 be comparable to the member reliability implied in current LRFD specifications (Buonopane
104 and Shafer, 2006; Zhang et al., 2014, 2016a,b). On the other hand, there are other types
105 of steel frames in which the system strengths predicted by elastic-LRFD and by inelastic
106 analysis are identical or close, frames in which static redundancies are limited or that are
107 redundant but fail elastically. System reliabilities and resistance factors for this category of
108 structures also must be investigated.

109 In this paper, we examine the system reliabilities of a number of simple yet representative
110 steel structures, including a continuous beam, a portal frame that fails elastically, a frame
111 with significant capacity for load redistribution, and two frames with limited capacity to
112 redistribute load. We compare the design strengths and system reliabilities of these structures
113 designed by LRFD and the inelastic method. Such a comparison can shed light on the system
114 reliability implications of LRFD and the inelastic method, and the suitability of the current
115 resistance factor stipulated in Appendix 1 of *AISC Specification 360-10*.

116 DESIGN BY INELASTIC METHOD

117 Probability-based limit state design criteria have the general format:

$$118 \text{Design strength} > \text{Required strength.} \quad (1)$$

119 In a system-based safety check, the required strength for a frame is defined by the loads
120 applied to the complete frame, i.e.,

$$121 \text{Required strength} = \sum \gamma_i Q_{ni}, \quad (2)$$

122 in which Q_{ni} are nominal loads, and γ_i are the load factors determined from the load com-
123 bination rules specified in loading standards (e.g., *ASCE Standard 7-10* (ASCE, 2010)). We

124 assume that the current load combination rules, which were developed for member-based
125 safety checks, are equally applicable to system-based safety checks.

126 When conducting an inelastic analysis for design, the loads are applied incrementally
127 to push down or push over the frame, depending on whether the system capacity under
128 gravity loads or gravity plus lateral (wind) loads is sought. In either case, the actual applied
129 loads are the product of the full load combination and an applied load ratio λ . The general
130 procedure for checking the integrity of a structural system by the inelastic design method in
131 Appendix 1 of *AISC 360-10* can be summarized as:

- 132 1. Develop an inelastic analysis model using the reduced nominal values of yield stress
133 $(0.9F_y)$ and modulus of elasticity $(0.9E)$ for all members;
- 134 2. Apply the loads $\lambda \sum \gamma_i Q_{ni}$ by increasing the applied load ratio λ incrementally until
135 collapse of the frame;
- 136 3. Check if the ultimate load ratio $\lambda_u \geq 1$.

137 In design by the inelastic method, the system strength limit state is characterized by frame
138 collapse/instability. The design strength of a frame is defined as the peak load in the frame's
139 load-displacement response. If the load-displacement response does not have a descending
140 branch, it is assumed in this study that the ultimate strength is reached when the slope of
141 the load-displacement curve reduces to 5% of its initial value ([Ziemian et al., 1991](#)).

142 DESCRIPTIONS OF EXAMPLE FRAMES

143 Five planar structures are investigated in this paper. The first, shown in [Fig. 1](#), is a three-
144 span continuous beam. The yield stress and modulus of elasticity are 345 MPa and 200 GPa,
145 respectively. The second is a portal frame, shown in [Fig. 2](#). The two columns are oriented
146 for minor-axis bending and the beam for major-axis bending. The steel has a nominal yield
147 stress of 345 MPa with a modulus of elasticity of 200 GPa. An initial out-of-plumbness of
148 1/500 is introduced. Cases 3, 4 and 5 are closely related two-bay two-story non-symmetric
149 frames, as shown in [Fig. 3](#). The three frames have the same layout (adopted from [Ziemian](#)

150 et al. (1991)). However, their member sizes and loads are different, as given in Table 1. The
 151 nominal yield stress and modulus of elasticity are 320 MPa and 200 GPa, respectively. The
 152 three frames all have an initial out-of-plumbness of 1/500. In all cases, residual stresses are
 153 assumed to distribute according to the pattern suggested in Galambos and Ketter (1959).
 154 All five structures are subjected to gravity load only.

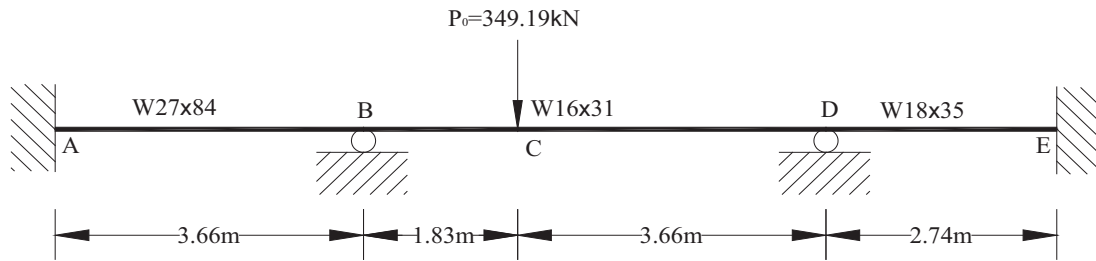


FIG. 1: Case 1: a three-span continuous beam.

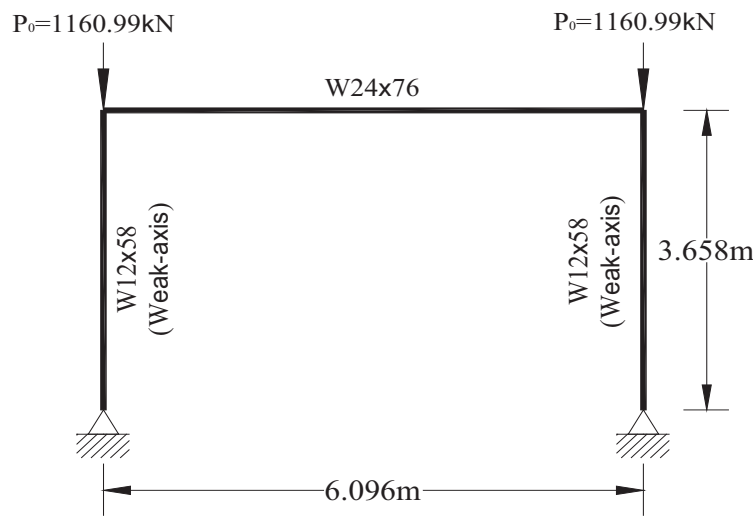


FIG. 2: Frame 2: a portal frame.

155 DESIGN STRENGTHS BY LRFD AND INELASTIC METHODS

156 For each structure, the design strength is determined using two methods, i.e., the elastic
 157 LRFD method and the inelastic method as it appears in Appendix 1 of the *AISC Specification 360-10*.
 158 With the inelastic method, the applied load ratio is increased until the frame
 159 collapses. The ultimate load ratio is denoted by λ_{in} . The inelastic analyses were performed

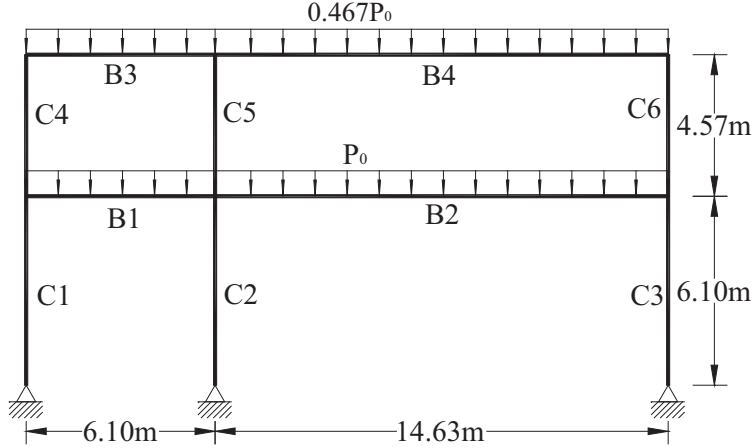


FIG. 3: Frame 3, 4 and 5: two-bay two-story frame.

TABLE 1: Member sizes and loads for Frame 3, 4 and 5.

	Frame 3	Frame 4	Frame 5
C_1	W12×19	W6×15	W6×15
C_2	W14×159	W14×99	W14×68
C_3	W14×145	W14×68	W14×68
C_4	W6×9	W6×8.5	W6×8.5
C_5	W14×145	W14×145	W14×145
C_6	W14×145	W14×145	W14×145
B_1	W30×116	W30×124	W30×132
B_2	W36×182	W36×182	W36×182
B_3	W24×55	W24×55	W24×55
B_4	W30×116	W30×116	W30×116
Loads (P_0)	146.93 kN/m	142.49 kN/m	107.25 kN/m

160 using the software package OpenSEES (Mazzoni et al., 2007). In the case of LRFD, the
 161 applied load ratio is increased until one of the members reaches its LRFD limit state, i.e.,
 162 the left-hand side of the beam-column interaction equation of Chapter H in the *AISC Spec-*
 163 *ification 360 -10* is equal to 1.0. The applied load ratio corresponding to the LRFD limit is
 164 denoted by λ_e . All structures considered are code-compliant.

165 The design strength of the three-span continuous beam in Case 1 is $\lambda_e = 1.0$ by LRFD
 166 and $\lambda_{in} = 1.29$ for the inelastic method. Fig. 4(a) plots the applied load ratio versus the
 167 vertical displacement at point C using the inelastic method. Fig. 4(b) shows the locations

168 of three plastic hinges developed at the inelastic ultimate limit point. As the inelastic
 169 analysis performed in this study is a plastic-zone type of inelastic analysis (as opposed to a
 170 plastic hinge type of analysis), it is assumed that a plastic hinge has developed if the cross-
 171 sectional yield ratio (percentage of cross-sectional area that has yielded) exceeds 75%. In
 172 this example, the design strength given by the inelastic method is 29% higher than LRFD.
 173 This is not surprising as the continuous beam has a significant capacity to redistribute loads
 174 after first yielding.

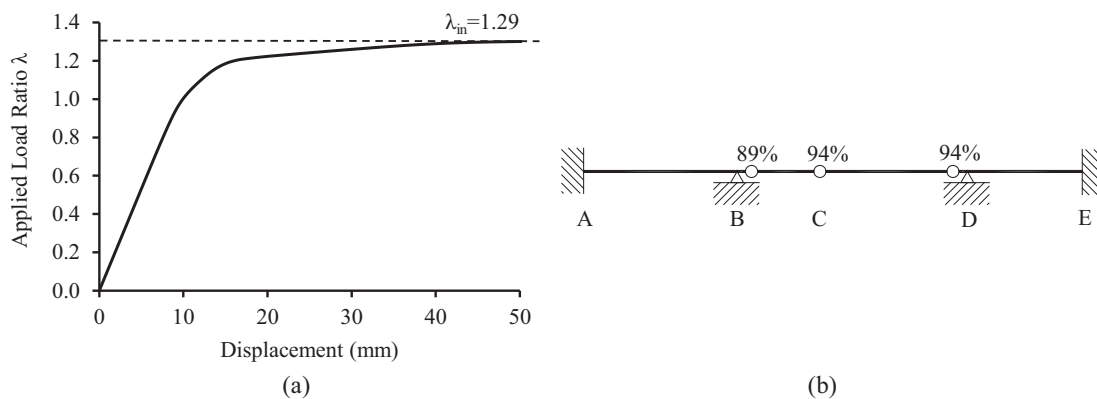


FIG. 4: Case 1: (a) load vs. displacement at C, (b) plastic hinges and section yield ratios at inelastic limit point.

175 A continuous beam can also be designed using the traditional plastic design method, in
 176 which a resistance factor of 0.90 is applied to the plastic moment capacities of all members.
 177 The design strength given by the plastic design method is $\lambda = 1.29$, identical to that of the
 178 inelastic method. This is to be expected as the traditional plastic design method is also
 179 based on a collapse limit state.

180 The design strength of the simple portal frame in Case 2 is $\lambda_e = 1.0$ by LRFD, and
 181 $\lambda_{in} = 1.02$ by the inelastic method. Thus, the two design strengths are very close to each
 182 other. This is because this frame consists of slender members, and the frame fails essentially
 183 elastically. There are significant geometrically nonlinear effects, but no plasticity.

184 The frame in Case 3 developed significant plasticity at its inelastic ultimate limit state,
 185 as shown in Fig. 5(b). The load-roof displacement response from the inelastic analysis is

186 plotted in Fig. 5(a). The design strength of the frame is $\lambda_e = 1.0$ for LRFD and $\lambda_{in} = 1.19$
 187 for the inelastic method.

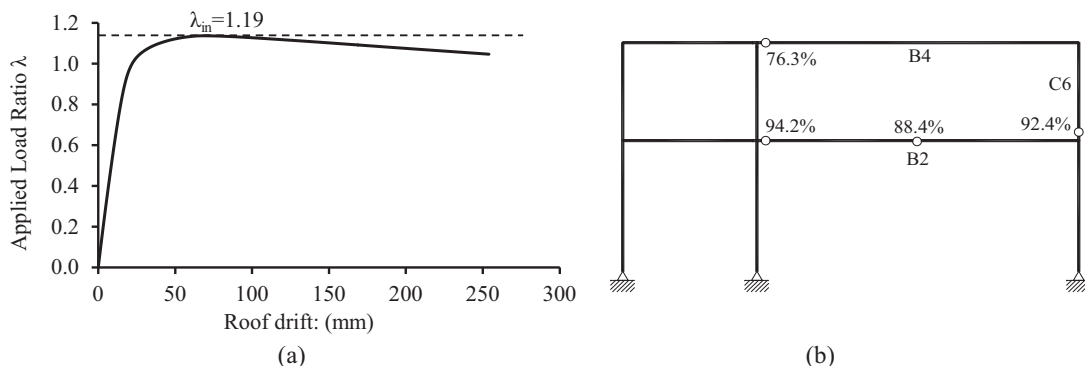


FIG. 5: Frame 3: (a) load vs. roof displacement, (b) plastic hinges and section yield ratios at inelastic limit point.

187
 188 Frame 4 represents a structure in which every beam and column is designed at or close
 189 to its LRFD-defined limit state. Fig. 6 shows the LRFD beam-column interaction equation
 190 (H1.1) calculations for every member of Frame 4, showing that the interaction equation
 191 calculations are all close to unity. The design strength of Frame 4 is found to be $\lambda_e = 1.0$ by
 192 LRFD, and $\lambda_{in} = 1.06$ by the inelastic method. The difference between the frame capacities
 193 determined by the LRFD and inelastic methods is only minor. The inelastic analysis shows
 194 that the frame has developed two plastic hinges at the ultimate limit point, one at the right
 195 end of beam B1 and one at the bottom of column C6. The two plastic hinges developed
 196 almost simultaneously when the applied load ratio λ was about 0.96. This frame has only
 197 a limited capability for redistributing forces after first yield. Fig. 7(a) and 7(b) show the
 198 load-roof displacement curve and locations of plastic hinges at the inelastic limit point.

199 Finally, Frame 5 represents a structure in which the strength is governed by one critical
 200 member. The design strength is $\lambda_e = 1.0$ for LRFD, and $\lambda_{in} = 1.08$ by the inelastic method.
 201 Since the frame has limited capability for load redistribution following the failure of the
 202 slender column C2, the system strength given by the inelastic method is only marginally
 203 higher than that given by LRFD. The inelastic analysis reveals that at the collapse limit,

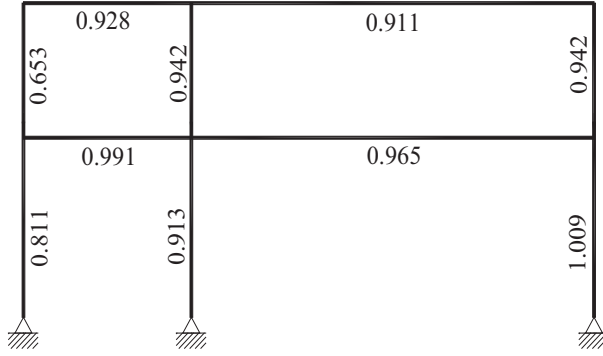


FIG. 6: Frame 4: LRFD beam-column interaction equation calculations for every member.

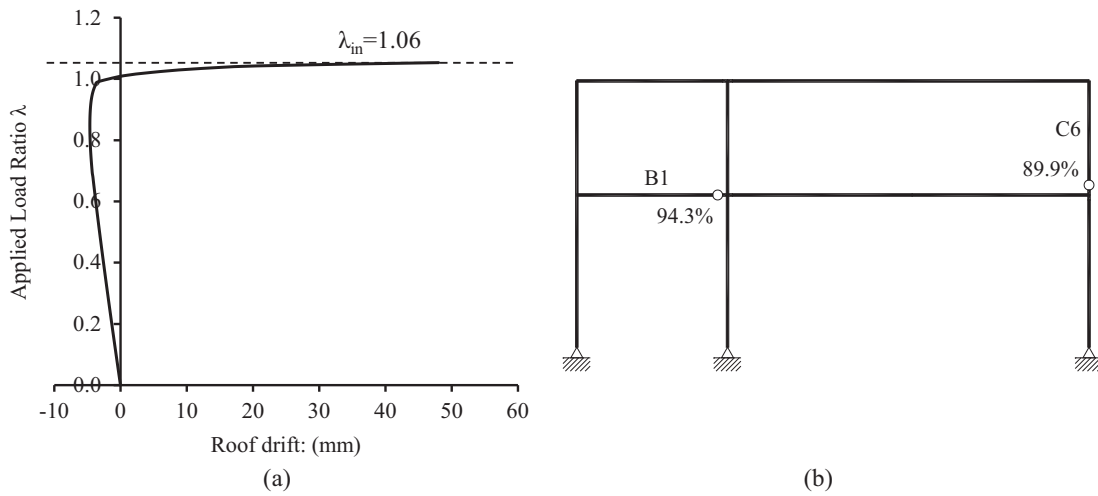


FIG. 7: Frame 4: (a) load vs. roof displacement, (b) plastic hinges and section yield ratios at inelastic limit point.

204 the slender column C2 is partially yielded with a yield ratio of about 58% (see Fig. 8(b)).
 205 Other members are all within their elastic limits at failure. Inelastic yielding only plays a
 206 minor role in this example, to a much lesser extent than in Case 1 and Case 3.

207 SYSTEM RELIABILITY ANALYSIS

208 Monte Carlo simulation

209 The collapse probabilities of the five structures designed by LRFD and inelastic methods
 210 were determined and compared. As shown in the previous section, the design strengths of a
 211 frame given by elastic LRFD and the inelastic method are generally different. The nominal
 212 load that a structure can support is $P_0 \cdot \lambda_e$ in the case of LRFD, and $P_0 \cdot \lambda_{in}$ in the case of

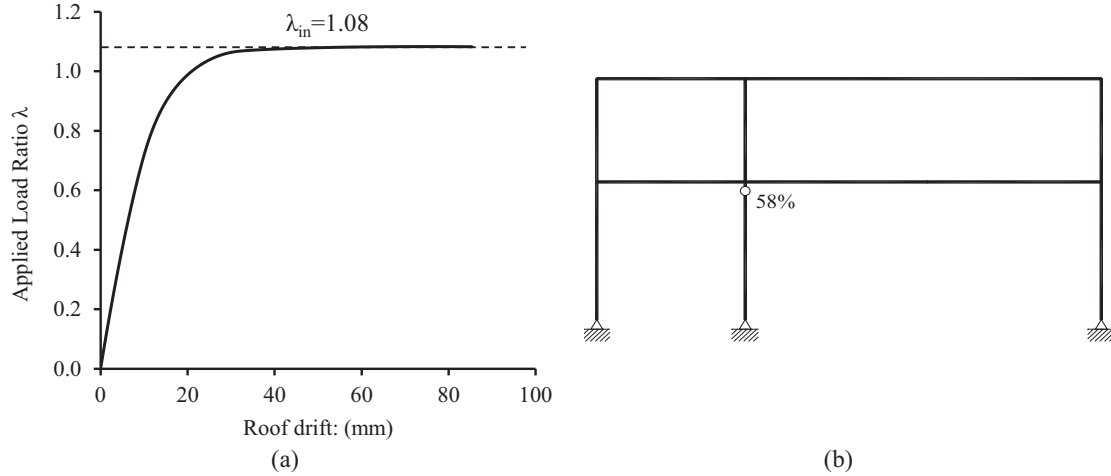


FIG. 8: Frame 5: (a) load vs. roof displacement, and (b) section yield ratio at inelastic limit point.

inelastic method, where P_0 is the reference load as shown in Figs. 1 to 3 and in Table 1. The structures are at their design limits in both design methods. Under this assumption we can meaningfully compare the reliabilities associated with both design methods.

The reliability assessments considered the following sources of uncertainties: live load and dead load, yield strength (F_y), Young's modulus (E), cross-sectional area (A) and moment of inertia (I) of each member. Table 2 summarizes the statistical information for these basic random variables. The properties among all columns, and among all beams are assumed to be perfectly correlated. However, the properties between beams and columns are uncorrelated.

The structural loads shown in Figs. 1 to 3 represent the gravity load combination $1.2D_n + 1.6L_n$, and the nominal live-to-dead load ratio is assumed to be $L_n/D_n = 1.5$. As the variability of live load is greater than that of dead load, a structure's reliability is dependent on the live-to-dead load ratio. Typical nominal load ratios L_n/D_n for steel structures vary from 0.5 to 5.0. The ratio 1.5 adopted in this study is approximately the weighted average of the nominal live-to-dead load ratios for steel structures (Ellingwood et al., 1982). This ratio, being greater than 1.0, reflects the fact that for steel structures, the live load tends to be greater than the dead load.

The system failure probability (for collapse), denoted by P_f , for each structure was eval-

TABLE 2: Description of basic random variables.

	$\frac{\text{mean}}{\text{nominal}}$	COV	Distribution	Ref.
D	1.05	0.1	Normal	Ellingwood et al. (1982)
L	1.0	0.25	T1Largest	Ellingwood et al. (1982)
F_y	1.10	0.06	Lognormal	Bartlett et al. (2003)
E	1.0	0.04	Lognormal	Bartlett et al. (2003)
A	1.0	0.05	Normal	Ellingwood et al. (1982)
I	1.0	0.05	Normal	Ellingwood et al. (1982)

COV = coefficient of variation

230 uated by More Carlo simulation. In each simulation, a random sample of the frame is
 231 generated with randomly generated values for material properties and cross-sectional prop-
 232 erties, according to their statistics in Table 2. Then an inelastic analysis was performed to
 233 determine if the frame would collapse under a random sample of the dead load and live load.
 234 After a sufficient number of simulations (20,000 to 10^6 simulations in this study depending
 235 on the value of P_f), the failure probability P_f is estimated by $P_f \approx n/N$, where N is the
 236 total number of simulations, and n is the number of simulations in which the frame collapsed
 237 (Melchers, 1999).

238 In applying structural reliability theory to structural code development, the probability
 239 of failure P_f is customarily converted to a reliability index β , which serves as an alternative
 240 and more familiar measure of reliability (Ellingwood, 1994). The reliability index is related
 241 to the probability of failure by $\beta = \Phi^{-1}(1 - P_f)$, in which $\Phi^{-1}(\cdot)$ represents the inverse
 242 function of the standard normal distribution function $\Phi(\cdot)$. In the following discussions, the
 243 system reliability index is denoted by β_s , in which the subscript s emphasizes it is a system
 244 reliability.

245 System reliability results

246 The system reliability indices for the five example structures designed by LRFD and the
 247 inelastic method are summarized in Table 3. Before we discuss the results of system reliability
 248 assessment, it should be noted that in the current LRFD of steel structures, the target

249 reliability index for compact rolled beams (plastic hinge limit state) and tension members
 250 (yielding limit state) is approximately 2.6-2.8 under the ASCE 7 gravity load combination
 251 (Ellingwood, 1994, 2000). In the following discussions, the term “member reliability” implies
 252 a reliability index about 2.6-2.8.

TABLE 3: System reliability indices and probabilities of failure for LRFD and inelastic method.

	LRFD		Inelastic	
	β_s	P_f	β_s	P_f
Beam	3.74	$9.201 \cdot 10^{-5}$	2.68	$3.681 \cdot 10^{-3}$
Frame 2	2.94	$1.641 \cdot 10^{-3}$	2.84	$2.256 \cdot 10^{-3}$
Frame 3	3.44	$2.909 \cdot 10^{-4}$	2.71	$3.364 \cdot 10^{-3}$
Frame 4	3.07	$1.070 \cdot 10^{-3}$	2.79	$2.635 \cdot 10^{-3}$
Frame 5	3.09	$1.001 \cdot 10^{-3}$	2.73	$3.167 \cdot 10^{-3}$

253 For the continuous beam, β_s is 3.74 for the elastic LRFD, 2.68 for the plastic design
 254 method, and 2.68 for the inelastic method. The plastic design method and the inelastic
 255 method lead to the same design strength, and thus the same system reliability. The prob-
 256 abilities of failure for elastic LRFD and the inelastic method differ by nearly two orders of
 257 magnitude difference (approximately $9.0 \cdot 10^{-5}$ vs $3.7 \cdot 10^{-3}$). A similar observation is made
 258 for Frame 3, in which β_s for LRFD is 3.44, significantly higher than that of the inelastic
 259 method ($\beta_s = 2.71$). Both the continuous beam and Frame 3 have significant capabilities for
 260 redistributing force following first yield.

261 Frame 2 shows a different characteristic in its system reliability than the continuous
 262 beam or Frame 3. Its system design strengths given by LRFD and the inelastic method are
 263 approximately the same, and thus its system reliability indices also are very similar; β_s is
 264 2.94 for LRFD and 2.84 for the inelastic method. This is to be expected, as Frame 2 fails
 265 essentially elastically with little yielding and force redistributing is not possible.

266 The system reliability β_s for Frame 4 is 3.07 ($P_f = 1.07 \cdot 10^{-3}$) if designed by LRFD, and
 267 2.79 ($P_f = 2.64 \cdot 10^{-3}$) if designed by the inelastic method. The two failure probabilities are

268 of the same order of magnitude. The system reliability of Frame 4, if designed elastically,
 269 is only marginally higher than the member reliability. This observation is different from
 270 Frame 3, although both are redundant frames. However, in Frame 4, all members are sized
 271 close to the LRFD limit state, and the frame has only limited capability for redistributing
 272 forces following first yield. In this case, the difference between system reliability and member
 273 reliability is only minor.

274 Finally, the reliability result of Frame 5 is similar to that of Frame 4; β_s is 3.09 for
 275 LRFD and 2.73 for the inelastic method. The two design methods lead to two system failure
 276 probabilities of the same order of magnitude. This is because Frame 5, like Frame 4, only has
 277 limited capability for load redistribution following the failure of the critical member (column
 278 C2).

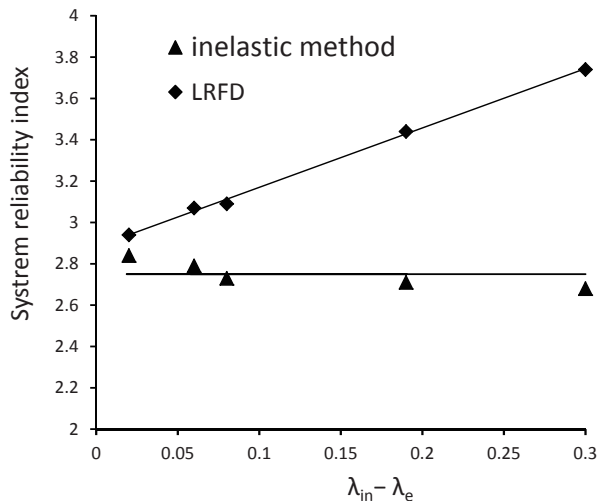


FIG. 9: System reliability index versus $\lambda_{in} - \lambda_e$.

279 Based on the system reliability analyses of these five structures, some general observa-
 280 tions can be made. If the frames are designed elastically to current practice, the system
 281 reliability indices are quite scattered; they vary from 2.94 (Frame 2) to 3.74 (the continu-
 282 ous beam), confirming the conventional wisdom that system reliability is generally higher
 283 than individual member reliability. The difference between system reliability and member
 284 reliability depends on the extent to which the structural system permits load redistribution

285 following first yield. This difference is greater for the structures with greater capability for
286 inelastic load redistribution, e.g., as seen in the continuous beam and Frame 3. Fig. 9 plots
287 β_s versus the difference, $\lambda_{in} - \lambda_e$, for the five structures, this difference represents the struc-
288 ture’s capability for inelastic load redistribution. It can be seen that β_s of the elastic LRFD
289 is approximately linearly proportional to $\lambda_{in} - \lambda_e$.

290 On the other hand, if the design is based on inelastic behavior, the β_s for different
291 structural systems are very consistent, ranging from 2.7 to 2.8 for the five frames as shown
292 in Fig. 9, despite their very different failure modes (elastic/inelastic) and structural behaviors
293 (capability for load redistribution). The consistency of the inelastic design method in meeting
294 system reliability goals can also be seen from Fig. 9. Similar observations have been made
295 elsewhere (Zhang et al., 2014). This study confirms that design by inelastic analysis is better
296 able to achieve uniform system reliabilities than the current member-based elastic methods.
297 Another point that is worthy of note is that the β_s for the inelastic method, being around
298 2.7-2.8, are about the same as the target member reliability index used for many of the
299 LRFD calibrations (Ellingwood, 1994, 2000). The reason that the inelastic method leads
300 to lower system reliabilities than LRFD is because the inelastic analysis takes much of the
301 reserve strength (after first yield) out of the system that elastic LRFD design leaves in.

302 TARGET SYSTEM RELIABILITY INDEX

303 By using the inelastic design method, it is possible to achieve a essentially uniform sys-
304 tem reliability for steel frames with different failure modes (elastic/inelastic) and behaviors
305 (capability for load redistribution). In particular, the inelastic design procedure in the Ap-
306 pendix 1 of *AISC Specification 360 -10* which applies a reduction factor of 0.90 to the yield
307 strength and stiffness of all members for the inelastic structural analysis, appears to achieve
308 a consistent system reliability index of 2.75 under gravity loads. This finding raises an im-
309 portant question: is a system reliability index of 2.75 sufficiently safe (or “acceptable”) for
310 a steel frame? In other words, is the reduction factor of 0.90 currently stipulated in the
311 Appendix 1 of *AISC 360* adequate?

312 The target reliabilities for the current *LRFD Specification* were obtained from a calibra-
313 tion procedure involving the reliability evaluations of a large number of members designed
314 using traditional acceptable practice (Ellingwood et al., 1982; Ellingwood, 1994). It would
315 be tempting to establish the target system reliabilities for inelastic design methods using
316 a similar calibration procedure. However, the range of system reliabilities implied by the
317 current *LRFD Specification* is quite wide. As shown in the previous section, the system
318 reliability indices for the four frames (Frames 2 to 5) designed by the current elastic-LRFD
319 practice span the range from 2.94 for Frame 2 which fails with virtually no material non-
320 linear action to 3.44 for Frame 3 which has a significant capability of load redistribution.
321 All these values of system reliability index are greater than the target member reliability
322 index originally stipulated for LRFD, which is approximately 2.6-2.8. On the other hand,
323 the system reliability index for the continuous beam is 2.68 if designed by the traditional
324 plastic design method, which has been an acceptable practice for many decades. Thus, at
325 least for continuous beams, a system reliability index as low as 2.7 would be acceptable.

326 The situation is more complex if frames, rather than continuous beams, are considered.
327 The selection of a target system reliability index should consider the mode and consequences
328 of structural failure (Ellingwood, 2001). The consequence of system collapse invariably is
329 more severe than the consequence of individual member failure in an indeterminate structure.

330 Furthermore, failures due to instability often occur more suddenly (especially if bifurca-
331 tion occurs) than failures due to yielding or plastic hinge formation, and may cause the entire
332 structure, or large portions of it, to collapse suddenly. If a distinction for mode/consequence
333 of failure is to be made (as suggested by Commentary Section C1.3.1 to *ASCE Standard*
334 *7-10*, which gives target reliabilities for different building Risk Categories), one could argue
335 that a higher target reliability should be stipulated for a system than for a member. On the
336 other hand, if the system target reliability is set too high, there will not be sufficient eco-
337 nomic incentives for engineers to adopt the more rigorous and performance-oriented inelastic
338 design method. A suitable target system reliability should be agreed upon by the stakehold-

339 ers in the building design process - professional engineers, researchers, and the regulatory
340 community (Ellingwood, 2001).

341 The results of this study suggest that the target system reliability index for steel struc-
342 tures designed by the Inelastic Analysis and Design method should fall in the range 3.0 to
343 3.25 under gravity loads. Under these conditions, the system failure probability would be
344 less than the member failure probability by a factor of approximately 2 to 5. To achieve this
345 reliability goal for systems designed by the inelastic method, the current resistance factor of
346 0.90 stipulated in the Appendix 1 of *AISC Specification 360-10* may need to be reduced to
347 the range 0.80-0.85.

348 **CONCLUSIONS**

349 Design by second-order inelastic analysis represents a new system-based paradigm for
350 steel design. This paper compares the strengths and system reliabilities of five representative
351 structures designed by LRFD and the inelastic method. The five example structures exhibit
352 different failure modes and structural behaviors.

353 The continuous beam and Frame 3 both have significant capacity for inelastic load redis-
354 tribution. The system strengths predicted by the inelastic analysis are significantly higher
355 (more than 18%) than those from elastic LRFD. On the other hand, Frame 2 fails with very
356 little inelastic action, and the system strengths given by LRFD and by inelastic method are
357 almost identical. Frame 4 and Frame 5 have limited capability of load redistribution, and
358 the difference in system strengths given by LRFD and inelastic method is minor (within
359 8%).

360 While the current LRFD method was calibrated to achieve a uniform member-reliability,
361 the system reliabilities of frames designed by LRFD are quite variable, varying from 2.94
362 (Frame 2) to 3.44 (Frame 3). The difference between system reliability and member reliability
363 depends on the extent to which the frame can redistribute load after the initial formation of
364 plastic hinges.

365 The inelastic method, on the other hand, leads to consistent system reliabilities for dif-

366 ferent structures; the system reliability index is about 2.7 to 2.8 for all five structures in-
367 vestigated, despite their very different failure modes and structural behaviors. This result
368 demonstrates that system-based design by inelastic analysis is better able to achieve uniform
369 system reliabilities than the current member-based LRFD method. This is to be expected,
370 as the inelastic method is explicitly based on overall system behaviors.

371 The current inelastic design procedure in Appendix 1 of *AISC Specification 360-10* re-
372 duces the yield strength and stiffness of all members and connections by a factor of 0.9 for the
373 inelastic analysis. This procedure leads to a consistent system reliability index of about 2.75,
374 which is comparable to the target member reliability implied by the current LRFD method.
375 Considering that the consequence of system collapse is more severe than that of individual
376 member failure in an indeterminate structure, the authors believe that it might be desirable
377 to adopt a target system reliability index somewhat higher than the member reliability in-
378 dex, e.g., a target system reliability index in the range of 3.0 to 3.25. To achieve this level of
379 system reliability, the reduction factor for the yield strength and stiffness of all members may
380 need to be reduced from the current value of 0.90 to 0.85. However, the determination of
381 target system reliability index ultimately must be agreed upon by the professional, research,
382 and regulatory communities.

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